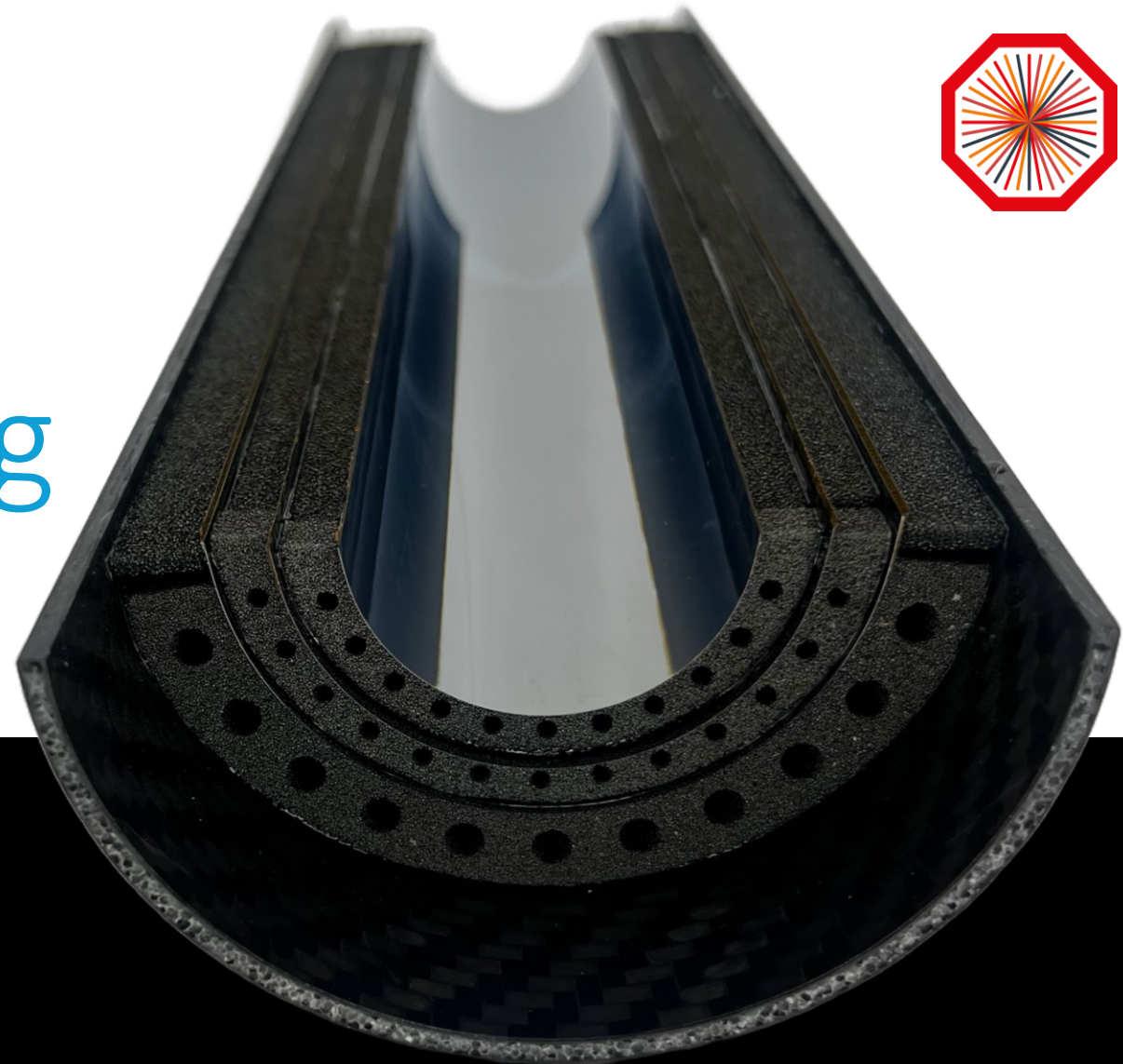


ALICE ITS3 vertex detector upgrade: mechanics & cooling overview



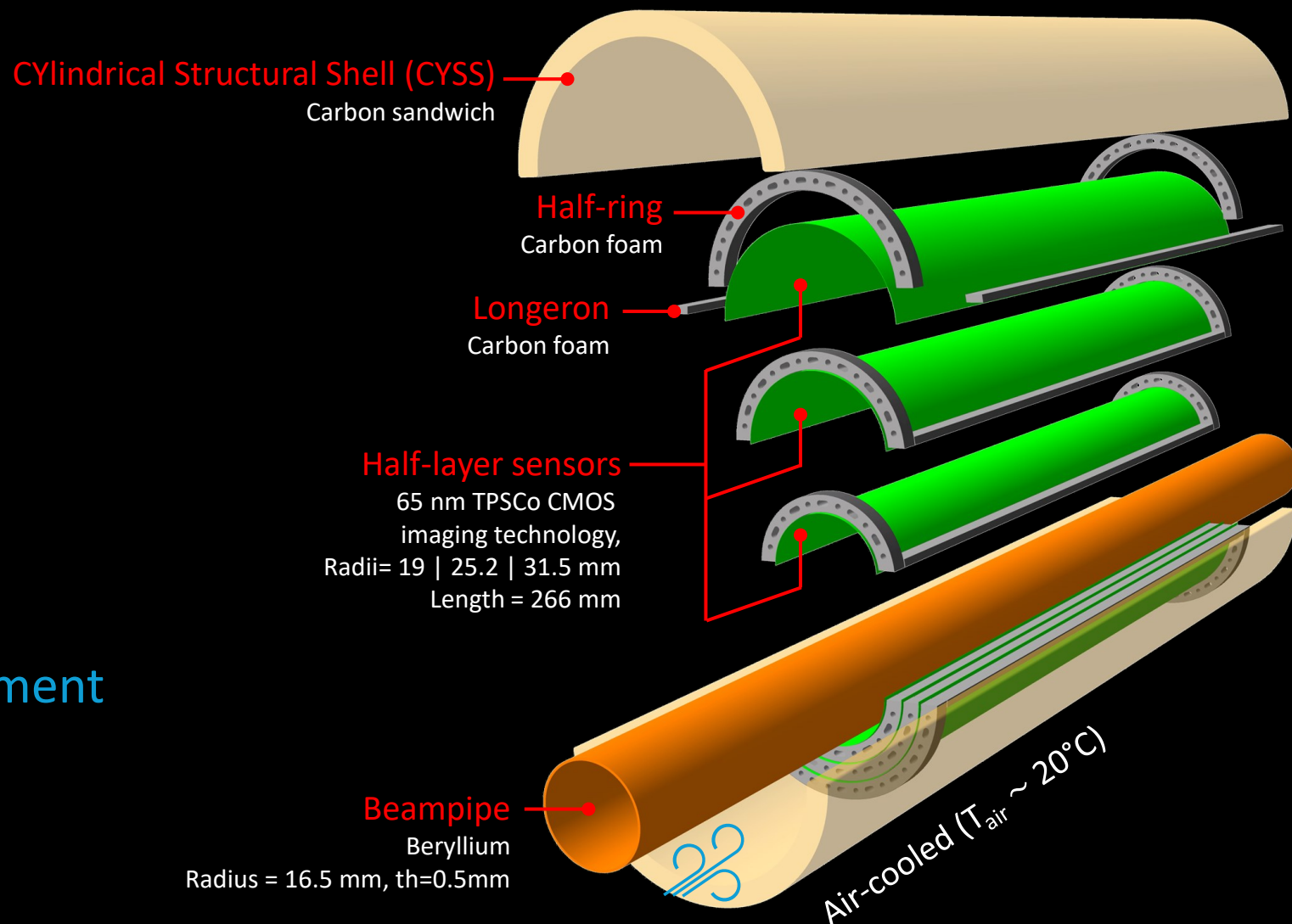
Corrado Gargiulo
on behalf of ALICE ITS3 Work Package 5



Simplified schematic of the ALICE Inner tracking system 3 (ITS3)

Outline

- Mechanics and Cooling
- Material budget
- Prototyping strategy
- Half-detector assembly
- Testing:
 - Cooling performance
 - Dynamic stability Vs airflow
 - Thermoelastic failure assessment
- Services

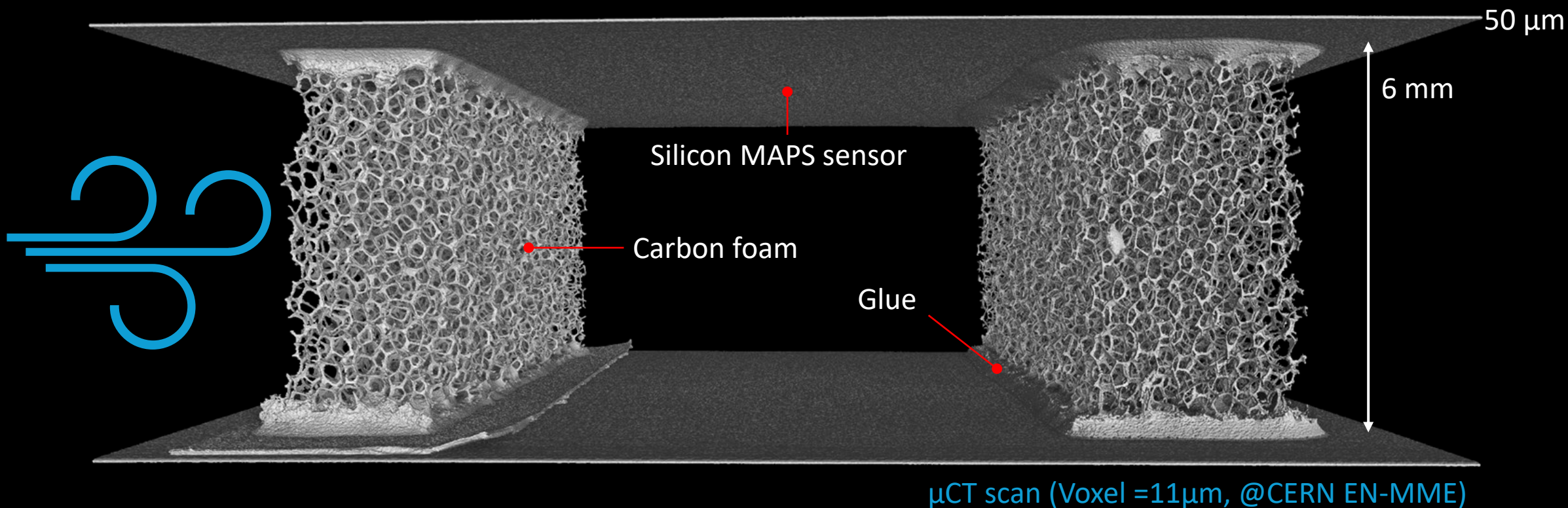


Mechanics and Cooling design



The limited dissipated power allows for the use of **air cooling** at ambient temperature

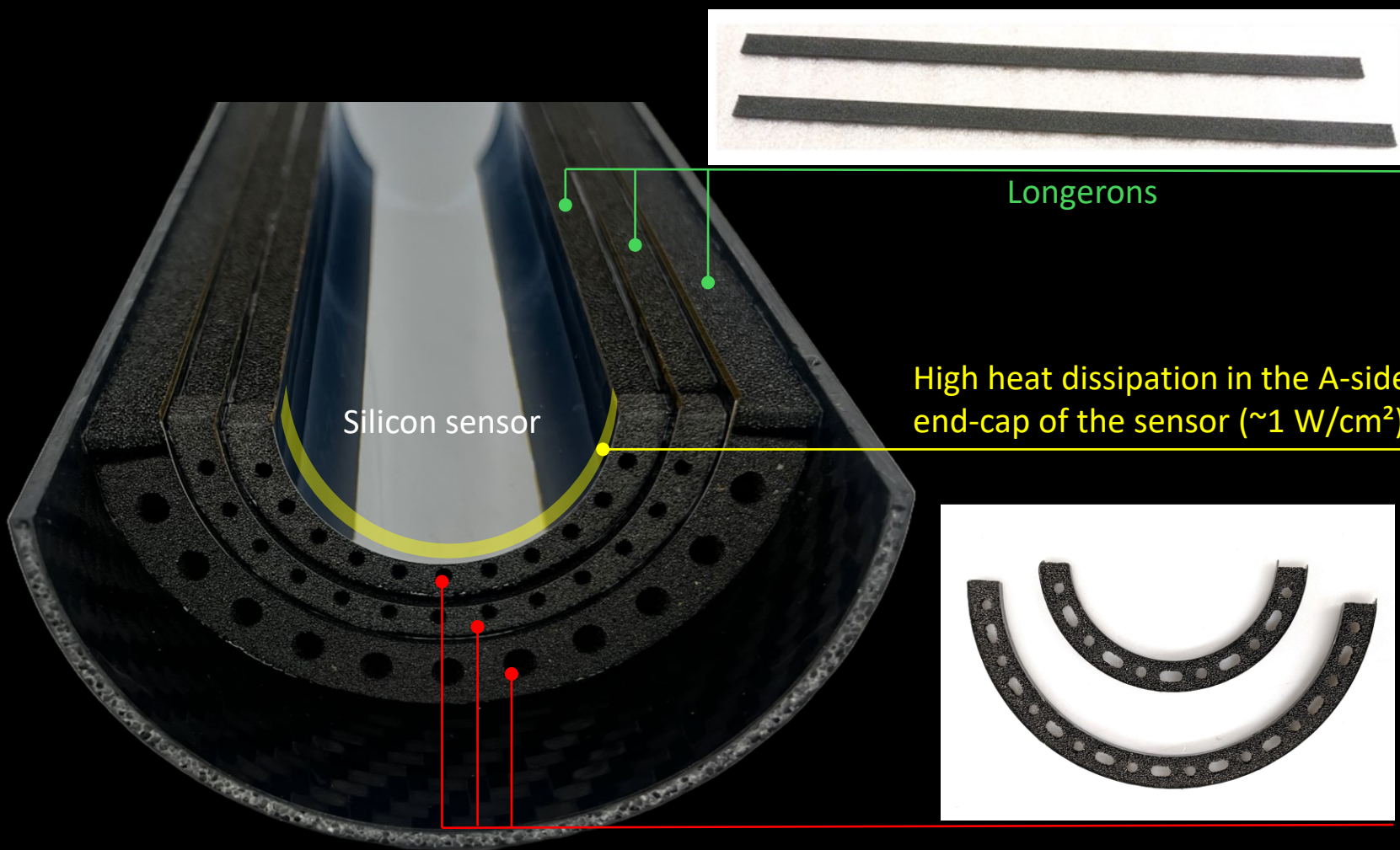
The material budget requirement (<1%) calls for an unpalpable support structure, i.e. **carbon foam** used as **support** and **radiator**



Mechanics and Cooling design



Power dissipation is concentrated at the short edge of the sensor, where carbon foam **half-rings** (radiators) are placed. Carbon foam **longerons** along the length keep the sensor in position and provide structural stability.



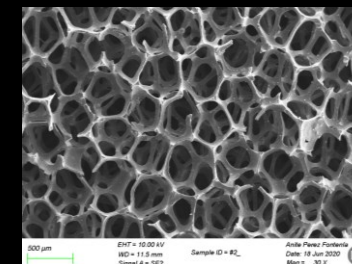
Longerons

High heat dissipation in the A-side end-cap of the sensor ($\sim 1 \text{ W/cm}^2$).

A-side half-rings

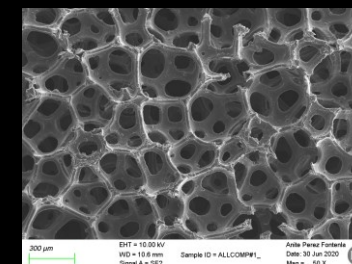
Carbon foam types

Support



ERG Duocel Carbon (RVC) Foam 100 PPI
 $\rho = 0.07 \text{ kg/dm}^3$
 $K = 0.033 \text{ W/m}\cdot\text{K}$

Cooling radiator

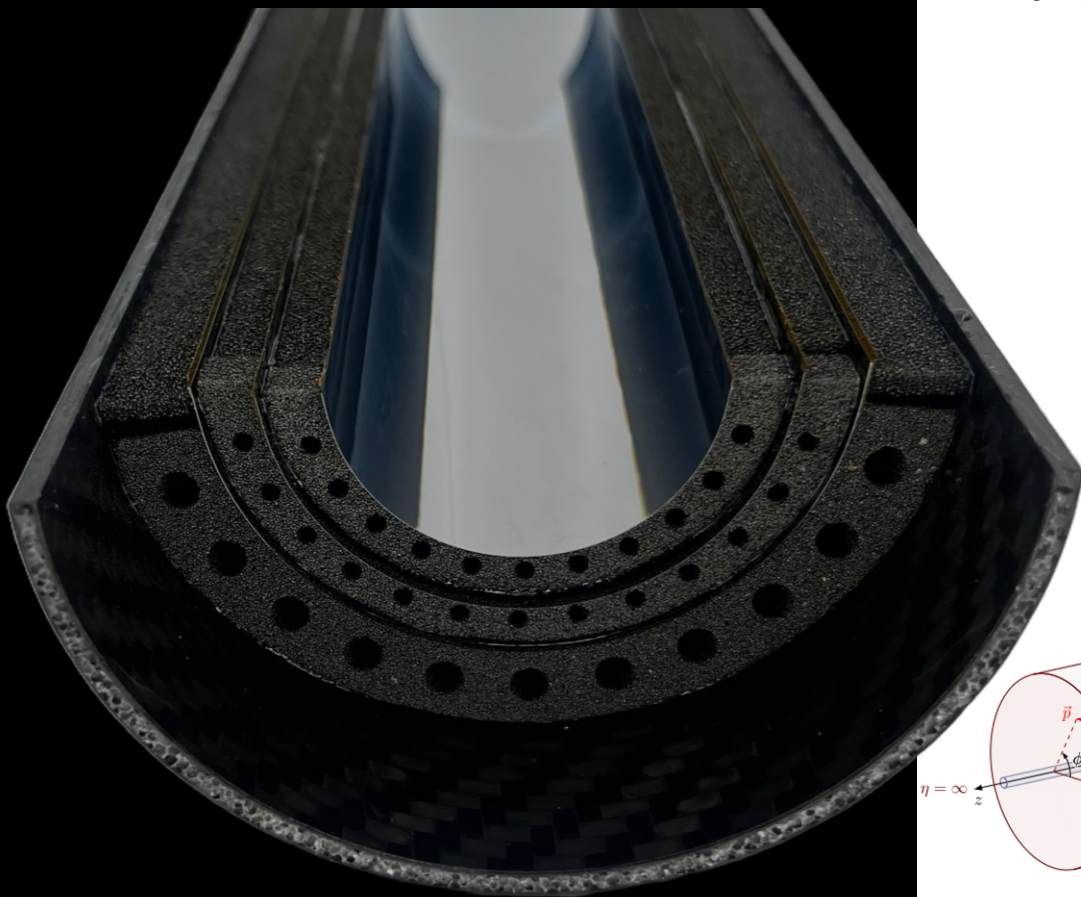


Allcomp K9 Standard Density (SD)
 $\rho = 0.2-0.26 \text{ kg/dm}^3$
 $K = >17 \text{ W/m}\cdot\text{K}$

Material budget

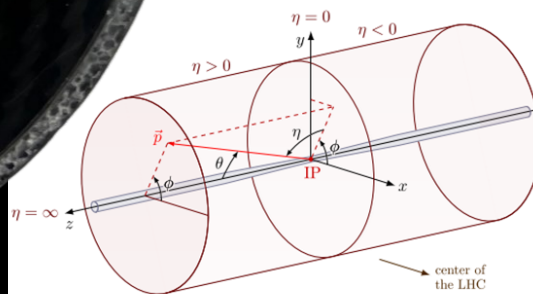
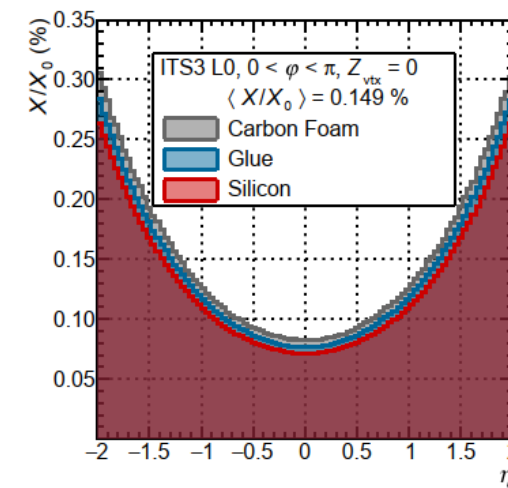
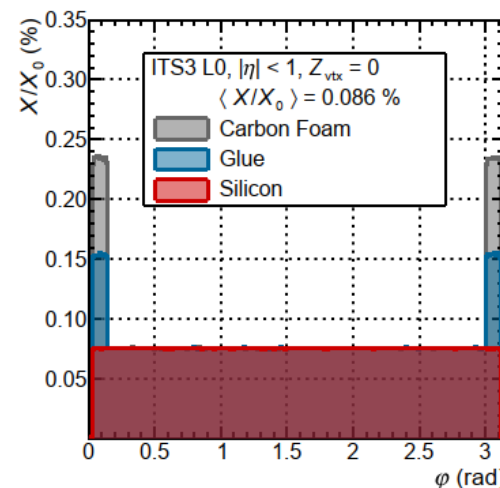


The half-layer layout has been developed to achieve **minimum material budget**, with most of the material budget belonging to the silicon sensor itself.



Material budget evaluation for half-layer 0

The silicon sensor itself is responsible for **0.07%** X_0 and the material budget for tracks with $|\eta| < 1$ on average is set at **0.09%** X_0 .



Material budget for tracks of particles originating from the interaction point ($Z_{vtx} = 0$) as a function of φ . The material budget plotted is averaged for tracks with $|\eta| < 1$, resulting in an average material budget contribution ($\langle X/X_0 \rangle$) of 0.086% for tracks with $Z_{vtx} = 0$, $|\eta| < 1$, and $0 < \varphi < \pi$.

Material budget for tracks of particles originating from $Z_{vtx} = 0$ as a function of η . The plotted material budget is averaged for tracks with $0 < \varphi < \pi$, resulting in $\langle X/X_0 \rangle = 0.149\%$ for tracks with $Z_{vtx} = 0$, $|\eta| < 2$, and $0 < \varphi < \pi$.

Courtesy of ITS3 WP1

Prototyping strategy



The strategy involves prototyping assemblies with varying levels of accuracy to validate the mechanics and cooling.

BreadBoard Models (BBMs):

Test samples and initial prototypes, partially representative of some of the final model features

Engineering Models (EMs):

Used for design development, they are a mixture of final-grade and commercial components

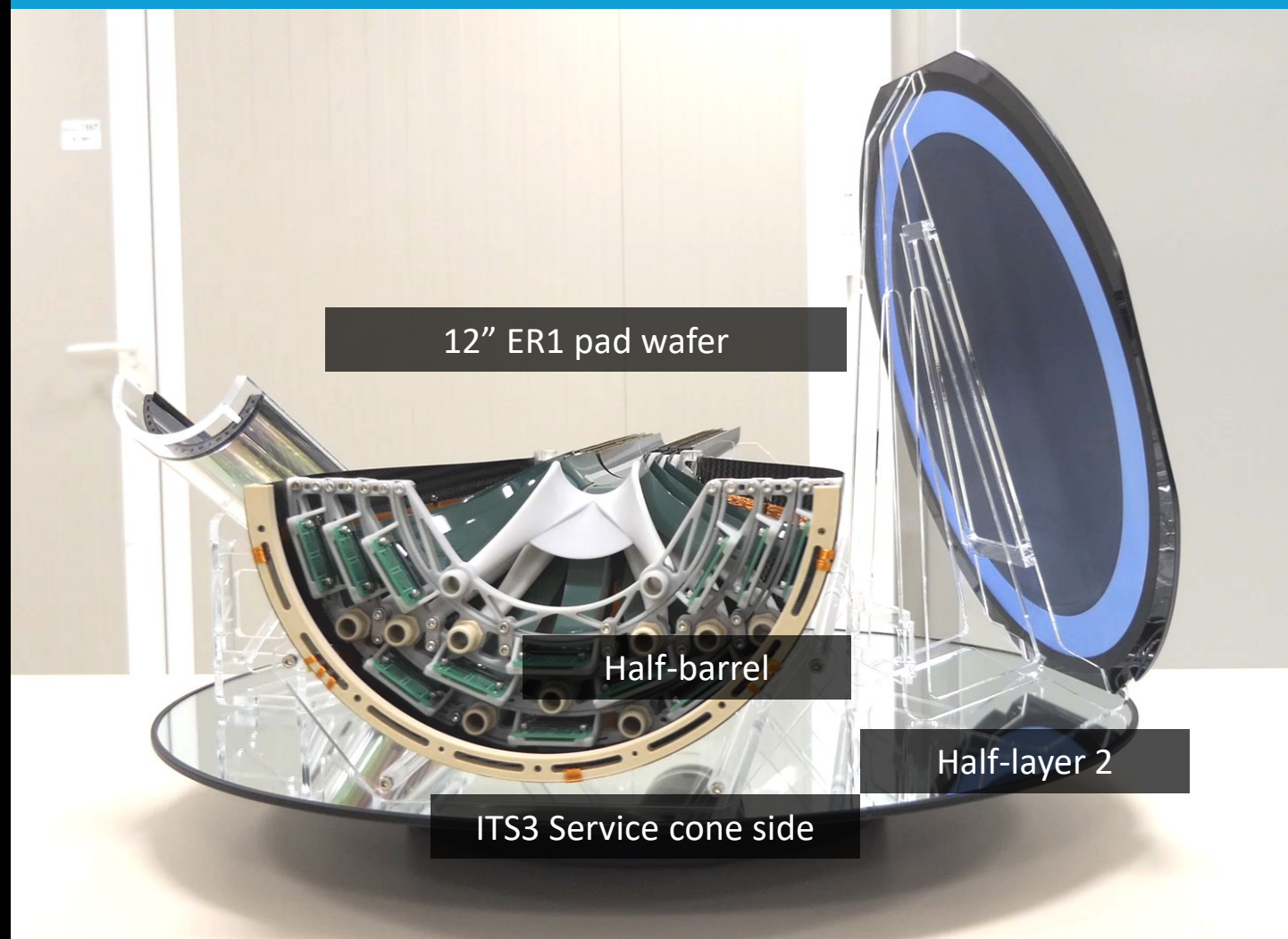
Qualification Models (QMs):

Final grade, fully integrated assemblies including MOSAIX sensors, used for qualification tests

Final Models (FMs):

2x final half-detectors to be integrated in the ALICE experiment
+ 2x half-detectors spares

ALICE ITS3 prototype



Prototyping strategy



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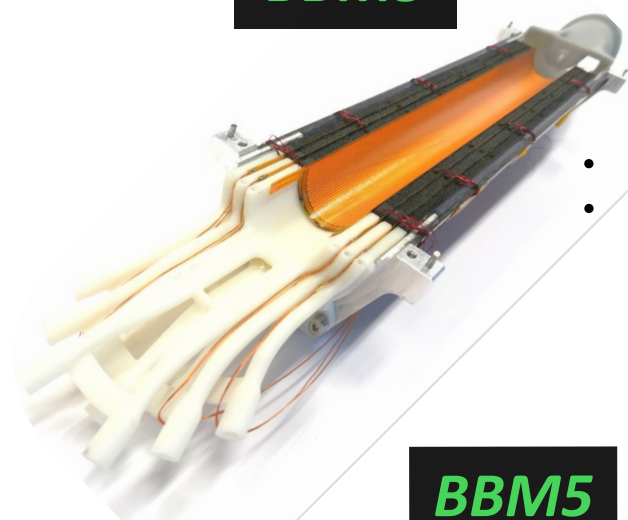
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Final Models (FMs):

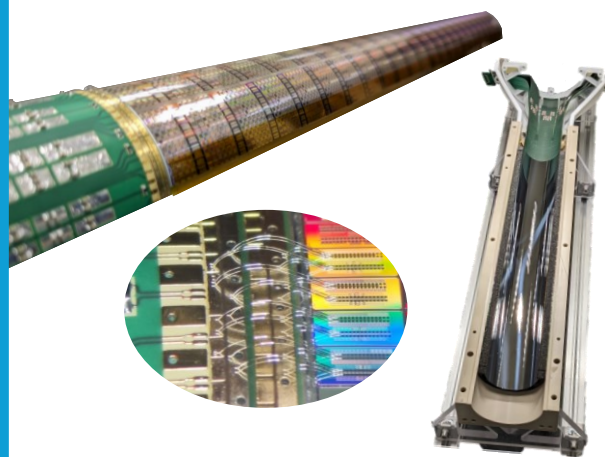
2x final half-detectors to be integrated in the ALICE experiment + 2x half-detectors spares



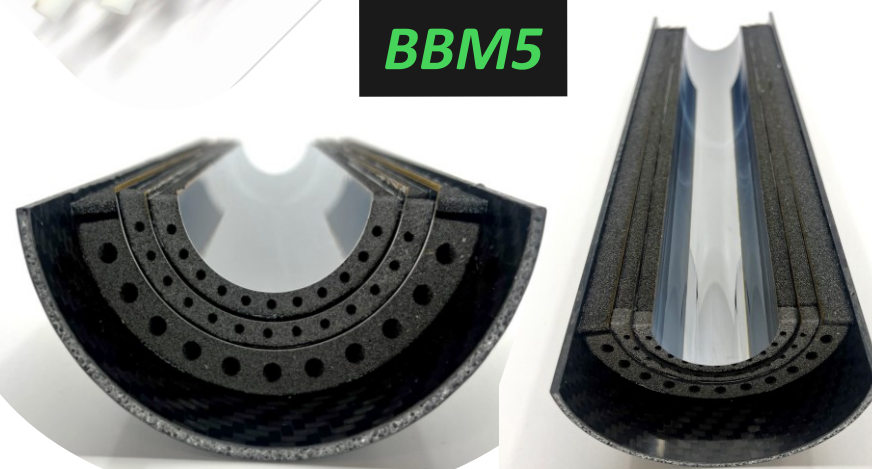
- Thermal testing



- Thermal testing,
- Aeroelastic testing



- (MLR1) wafer-size sensor integration,
- FPC-sensor wire bonding,
- Aeroelastic testing



- Thermoelastic testing

Prototyping strategy



The strategy involves prototyping assemblies with varying levels of accuracy to validate the mechanics and cooling.

BreadBoard Models (BBMs):

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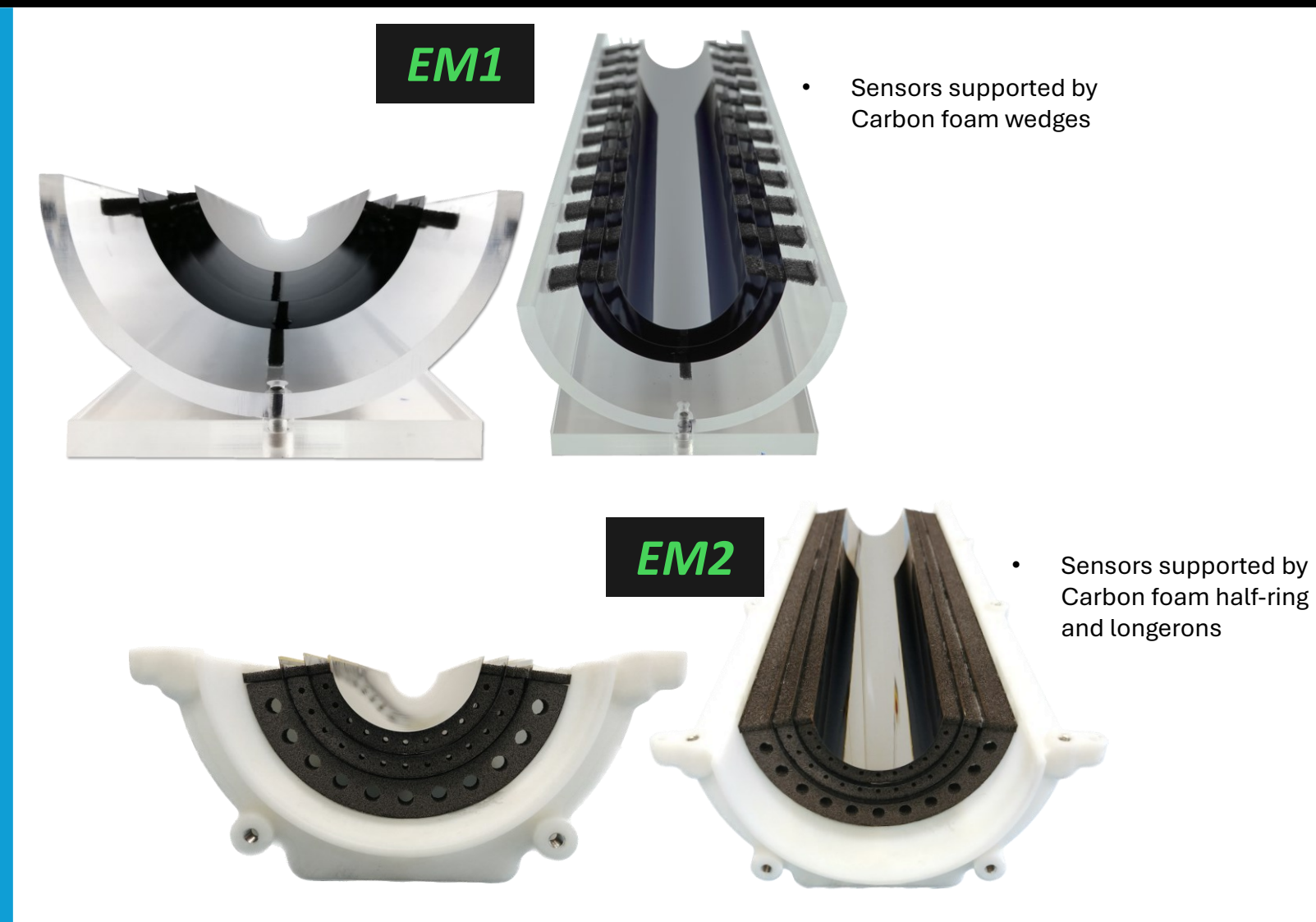
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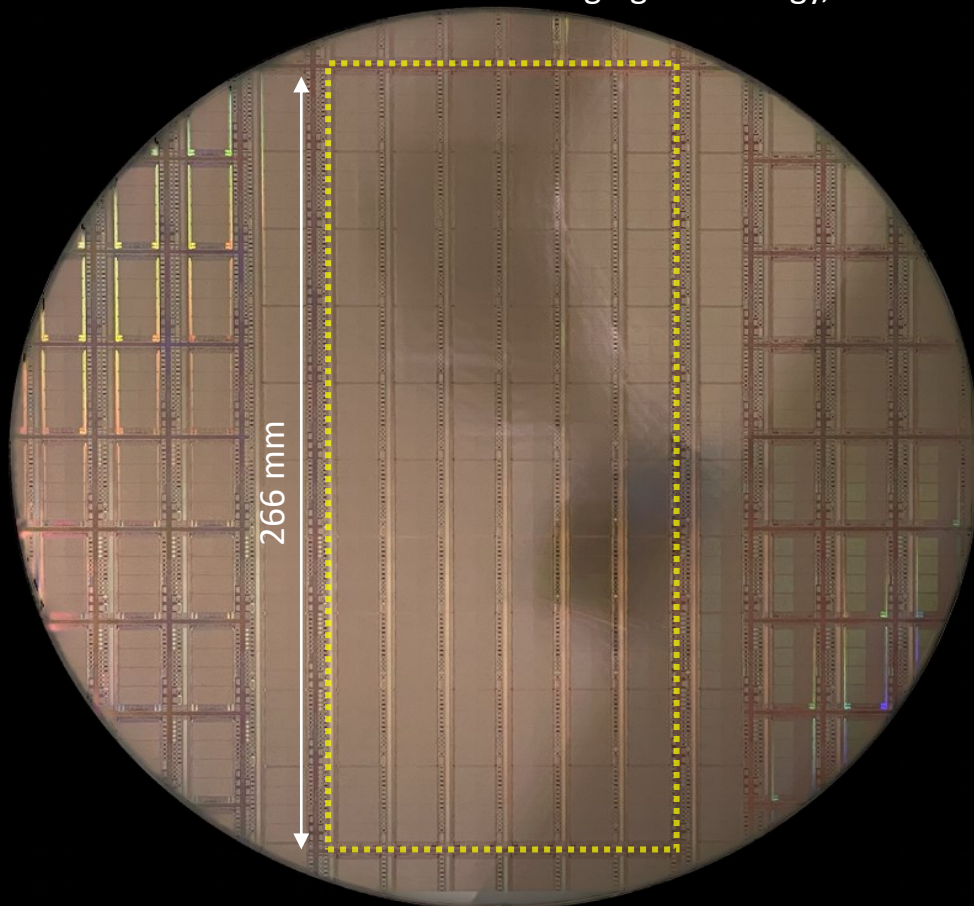
Detector assembly: Half-layer bending



Challenge: **bending** of the wafer-size thin silicon sensor without inducing stresses or failure

12" Si wafer, 50 μm

65 nm TPSCo CMOS imaging technology,

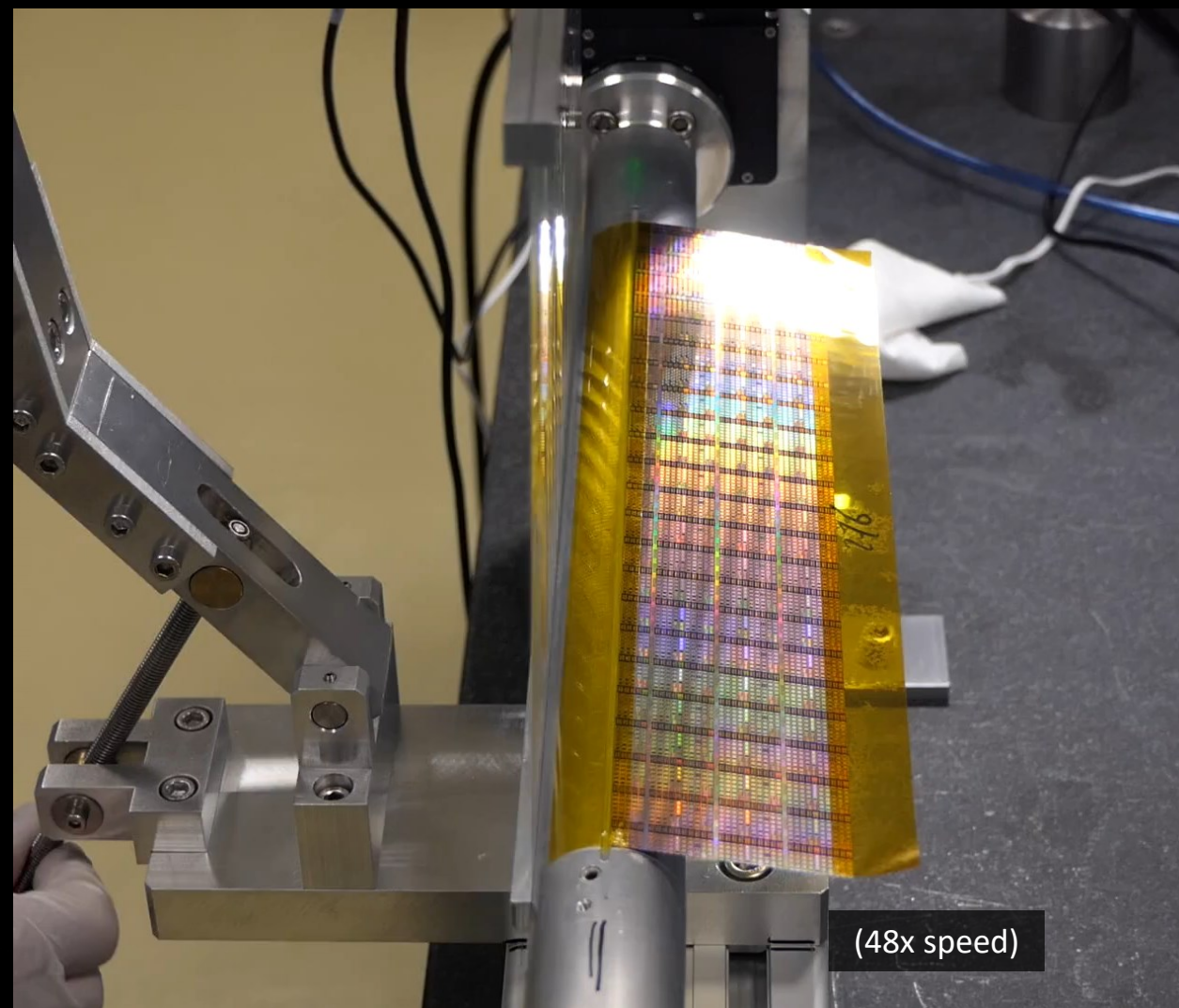


Half-layer 0 a = 58.7 mm

Half-layer 1 a = 78.3 mm

Half-layer 2 a = 97.8 mm

Bending procedure

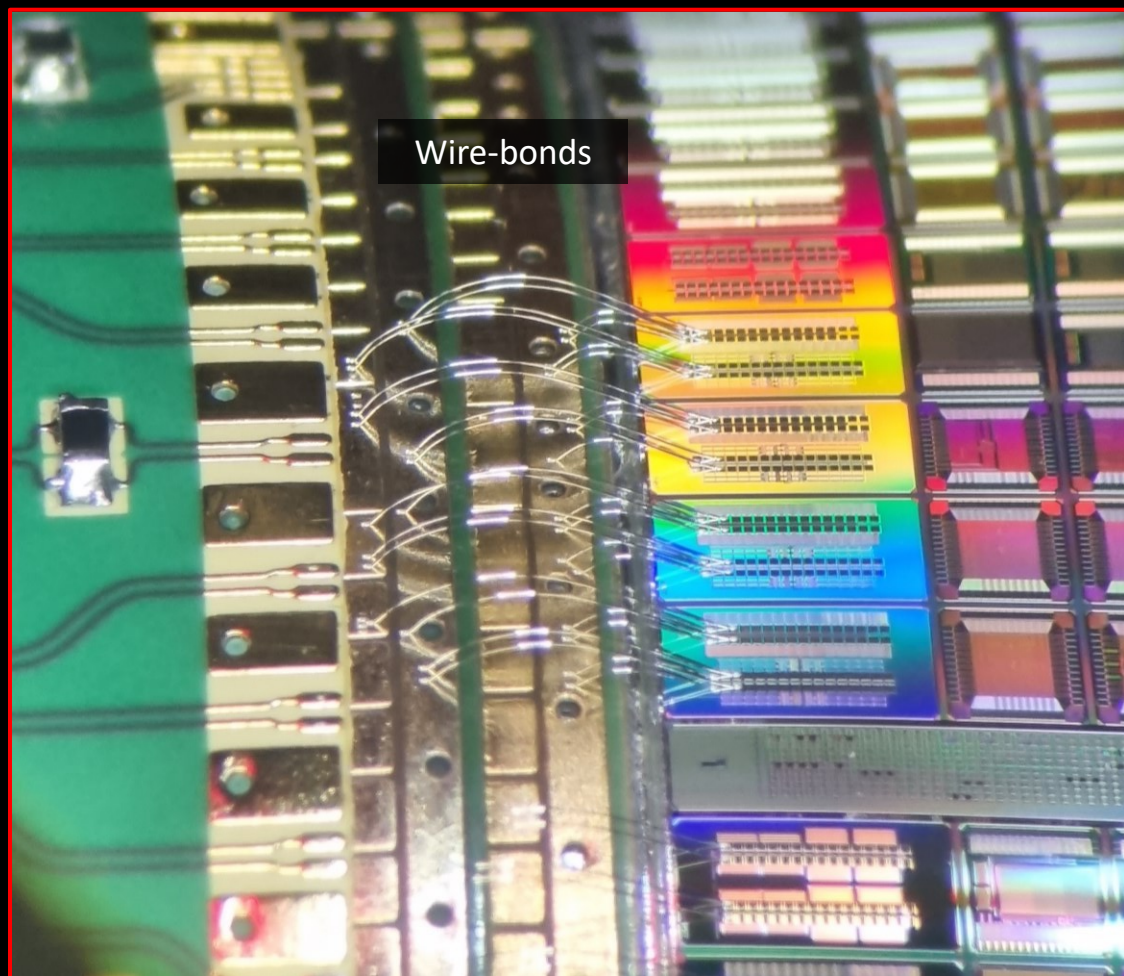


(48x speed)

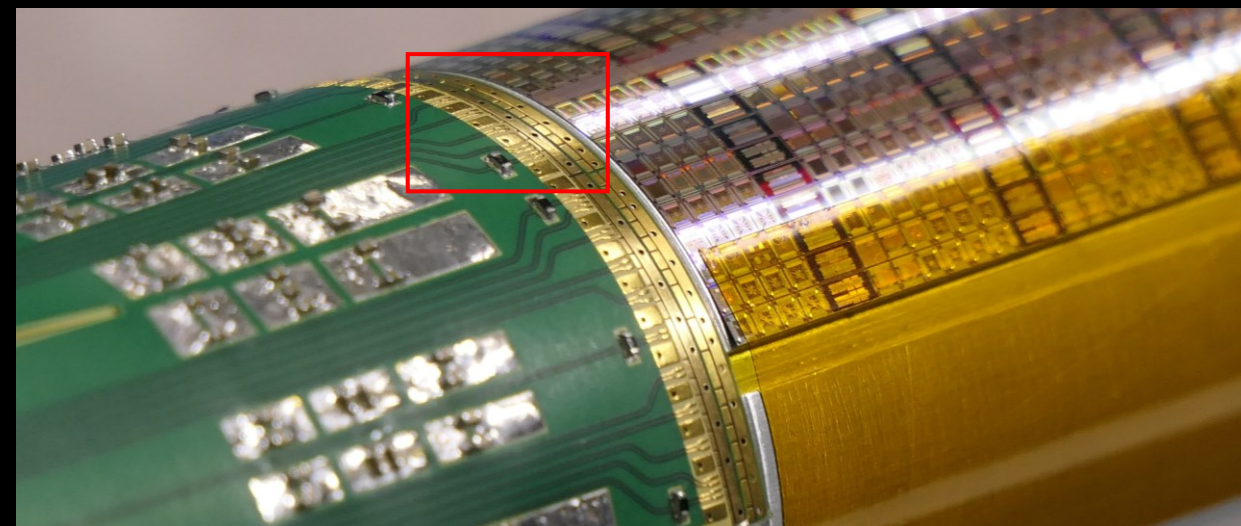
Detector assembly: Half-layer wire bonding



Challenge: is the electrically connection by **wire bonds** of the **curved sensor** to the close front end electronics



Electrical interconnection



Detector assembly: carbon foam Vs silicon gluing



Challenge: **optimum glue penetration** thickness (minimum material budget Vs thermal conductivity) in the foam, and a **smooth surface finishing**, avoiding punctual stresses and footprints.

- Carbon foam
- Glue penetration $\sim 200 \mu\text{m}$
- Carbon fleece $\sim 100 \mu\text{m}$
- 2nd Glue deposition $\sim 100 \mu\text{m}$
- Silicon sensor

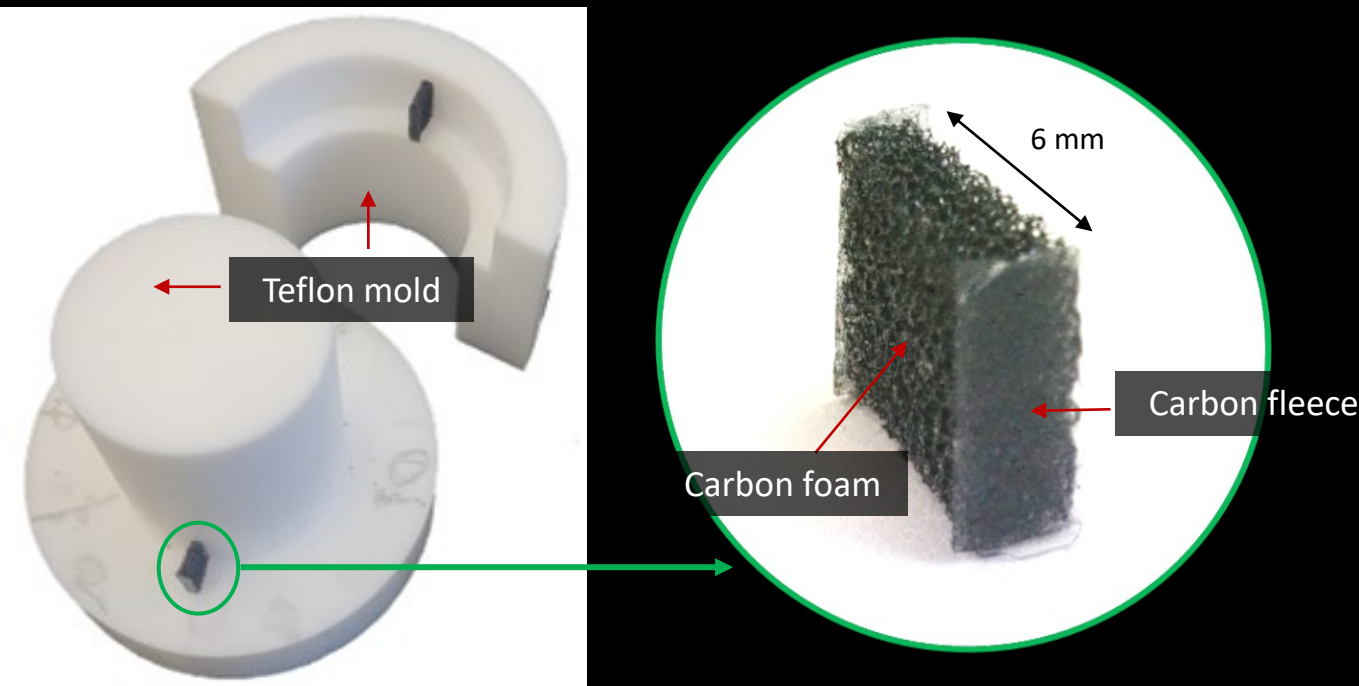


The carbon fleece veil is glued to the carbon foam surface in contact with the sensor and cured in a Teflon mold for a smooth finish.

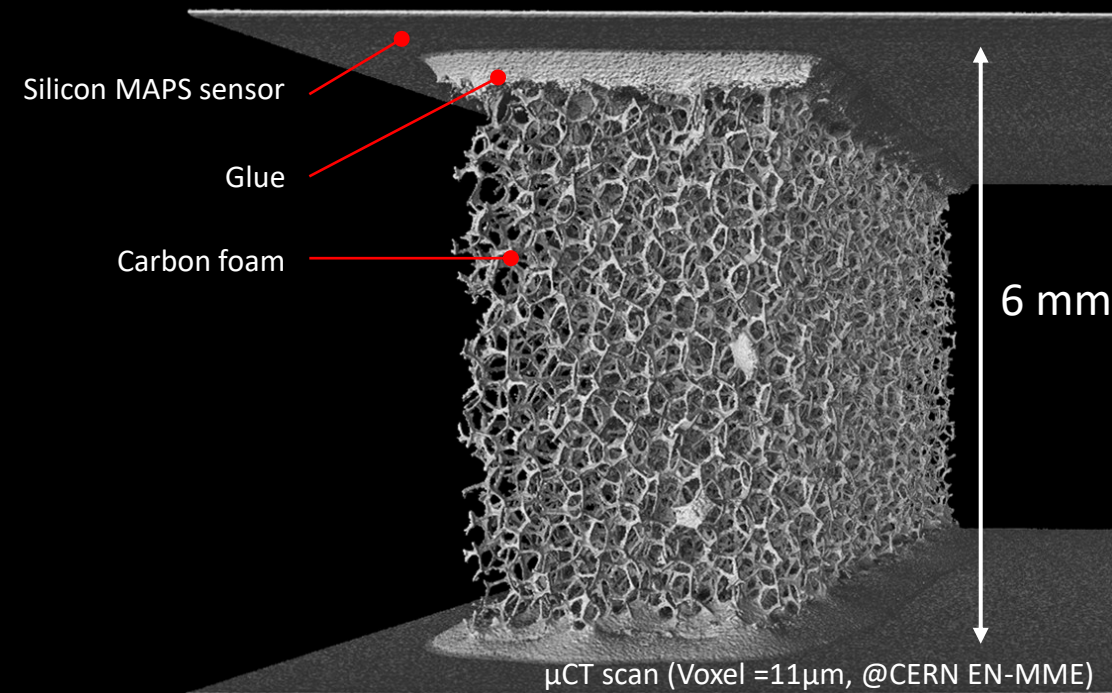


Subsequently, the carbon foam is glued to the silicon sensor, the glue thickness is 0.1 mm.

Step1 – Smooth surface finish



Step 2 – Gluing to Silicon

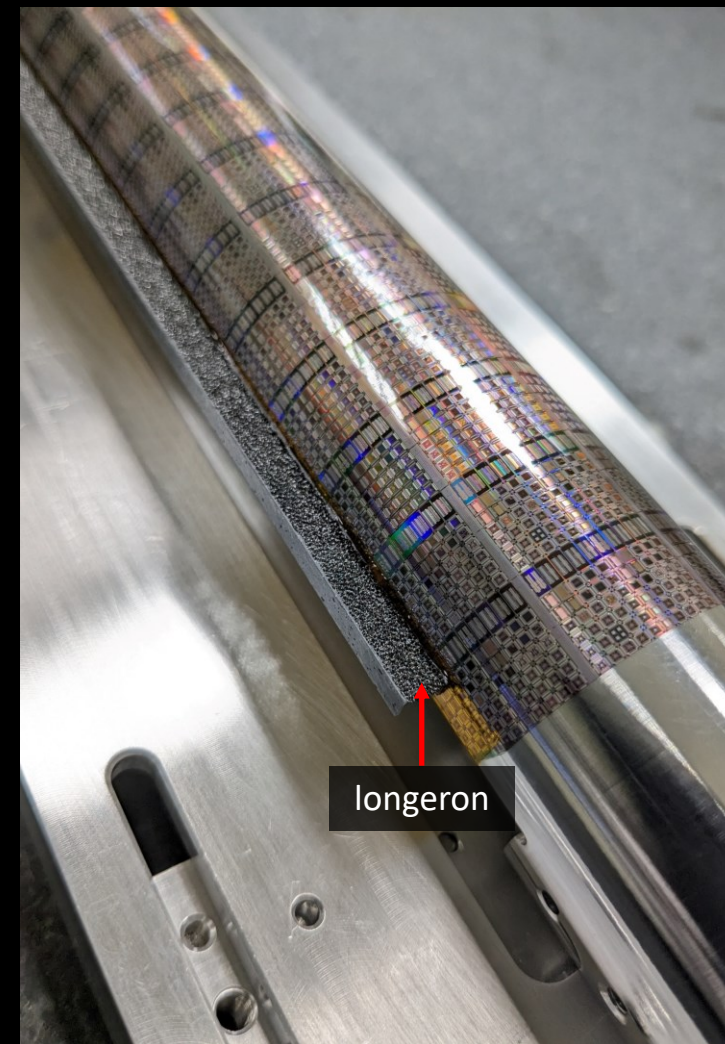
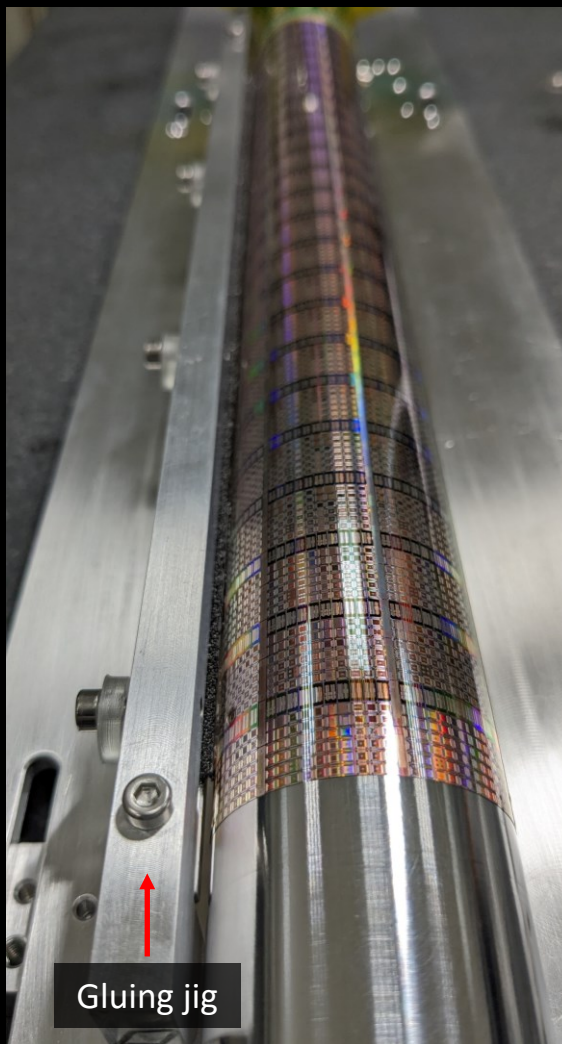


Detector assembly: Half-layer longerons



Challenge: **precise machining, positioning** and **gluing** of the carbon foam support

Gluing of the longerons



Detector assembly: Half-layer rings



Challenge: **precise machining, positioning** and **gluing** of the carbon foam support



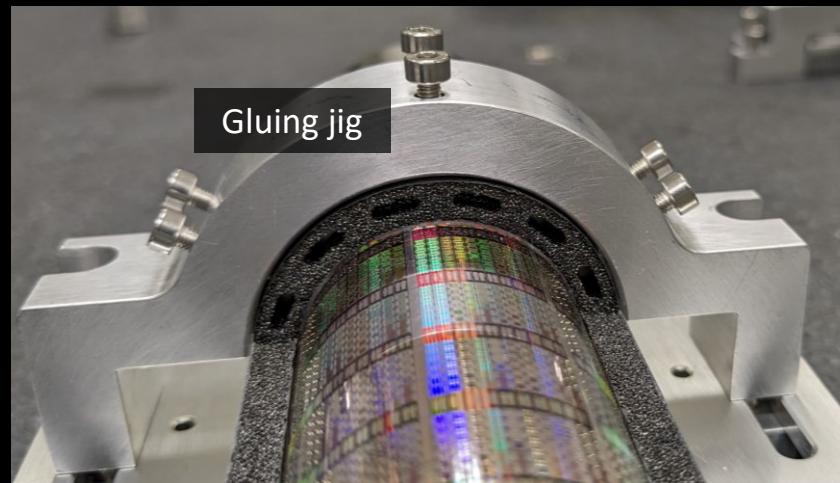
A-side half-rings:

Allcomp K9 SD

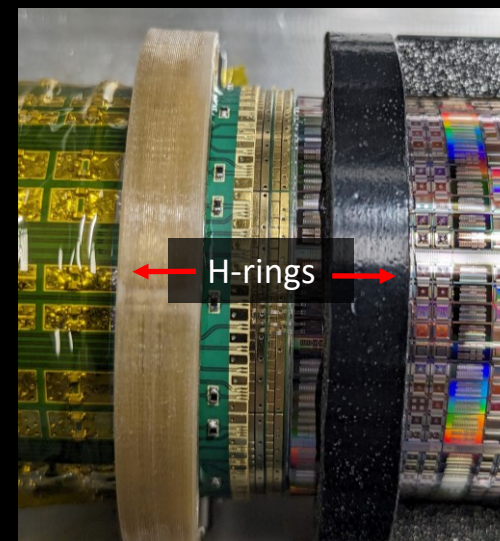
C-side half-rings:

ERG Duocel Carbon (RVC) foam

Gluing of the half-rings



Gluing jig



H-rings

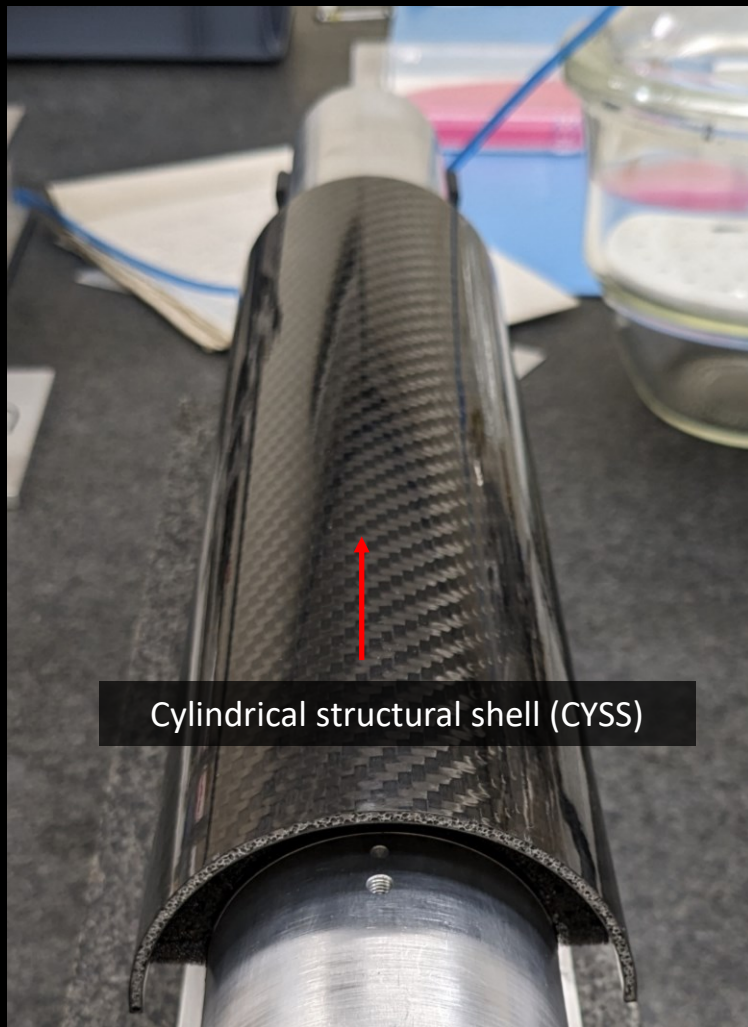


Detector assembly: Half-barrel



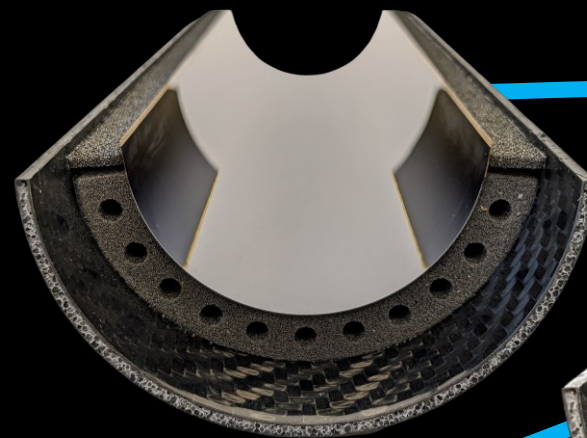
Challenge: minimum material budget achieved by a thin carbon cylindrical **exoskeleton** for the support to the three half-layers

H-L2 integration (Gluing deposition, alignment, curing)

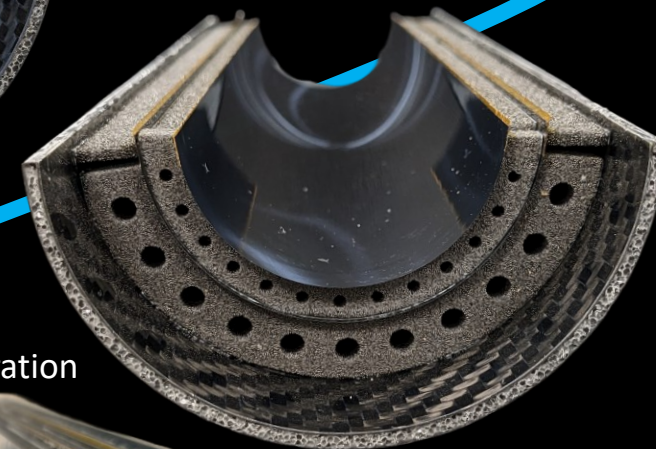


Half-detector after each half-layer integration

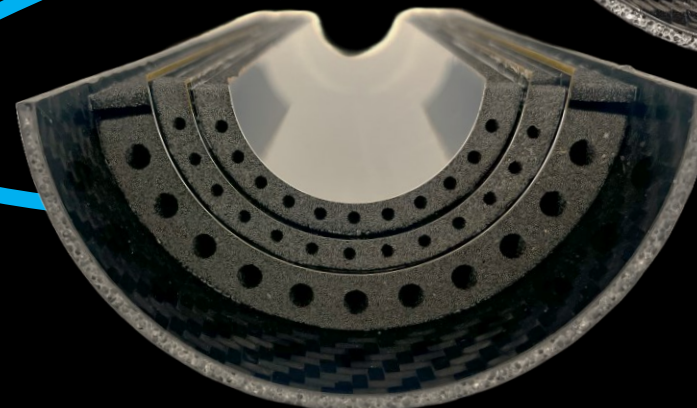
Half-layer 2 integration



Half-layer 1 integration



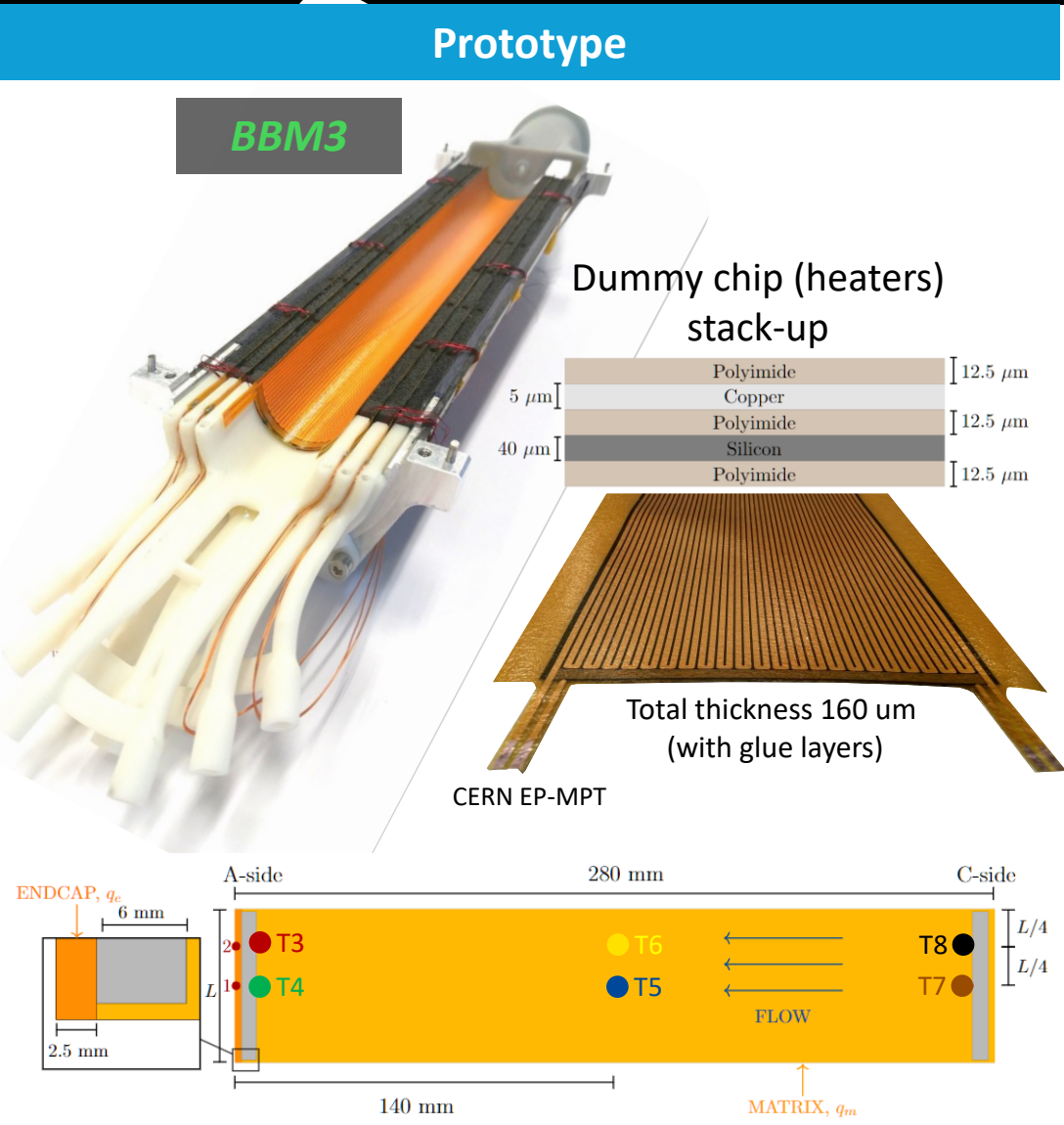
Half-layer 0 integration



Testing: Cooling performance



An airflow through the carbon foam radiator ring of **8m/sec** allows to keep the sensor below 25°C with an air inlet of 20°C



Experimental results

Two zones of different power dissipation:
 Endcap and Active area
 Same freestream velocity v_∞ in all layers, $v_\infty = 8 \text{ m/s}$
 Temperature of the inlet air $T_\infty \approx 20^\circ\text{C}$

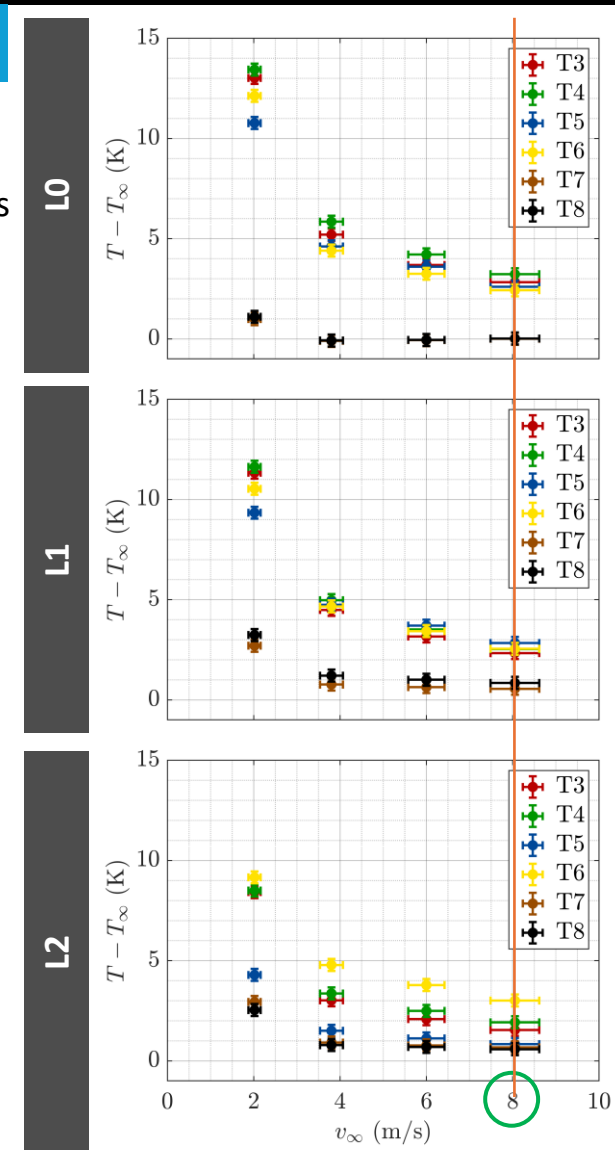
Surface power dissipation

Left End-cap: $q_e = 1000 \text{ mW/cm}^2$, uniform

Active area: $q_a = 50 \text{ mW/cm}^2$, uniform

Estimates of power consumption (TDR)

	Power density [mW cm ⁻²]		
	Expected 25 °C	Max 25 °C	Max 45 °C
Left End Cap (LEC)		791	
Active area (RSU)	28	44	62
Pixel matrix	15	32	51
Biasing	168	168	168
Readout peripheries	432	457	496
Data backbone	719	719	719



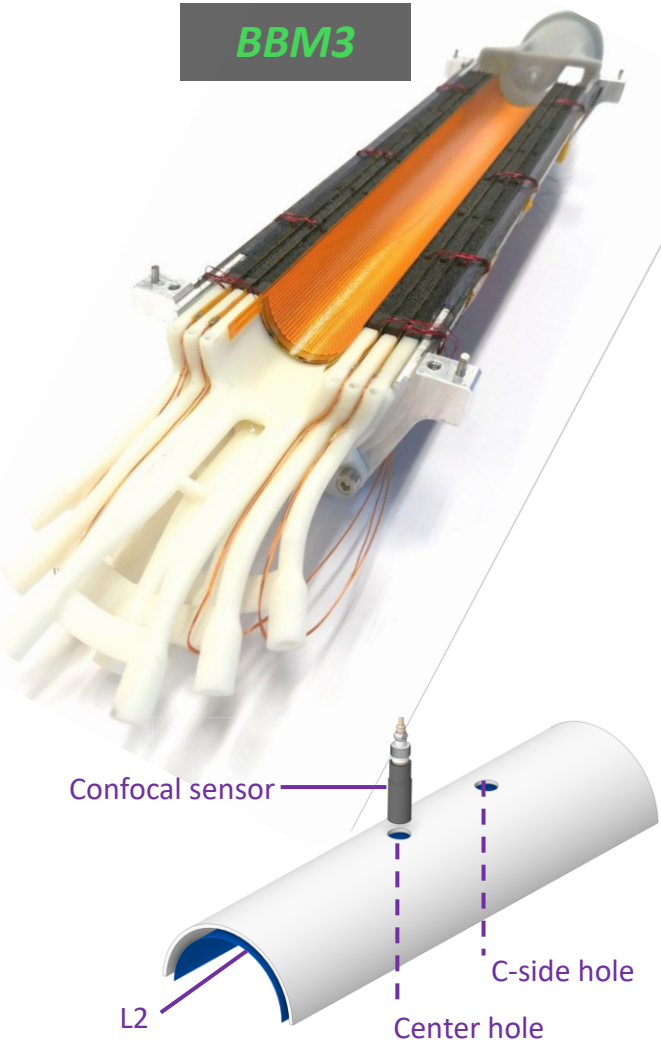
Testing: Dynamic stability Vs airflow



The experimental test results align with the simulation, showing a peak-to-peak displacement of approximately 1.1 μm .

Prototype

BBM3



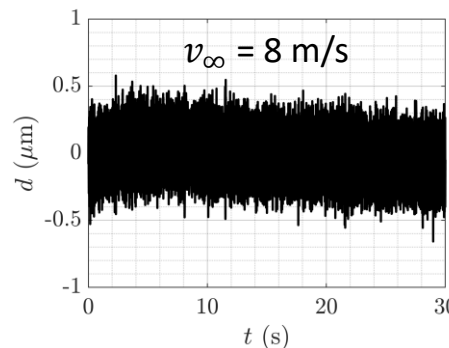
Experimental results

Confocal chromatic displacement sensor (at 30 kHz)
 Freestream velocity $v_\infty = 8 \text{ m/s}$

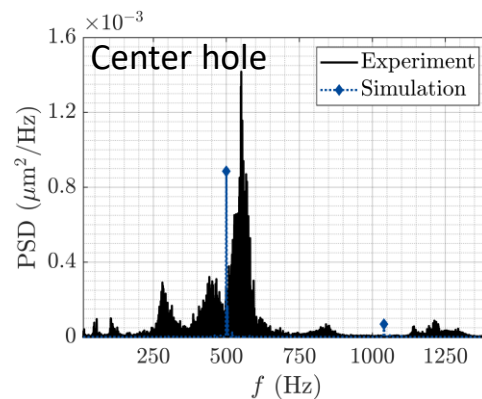
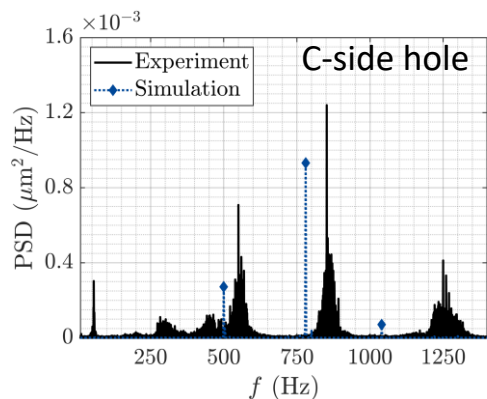
Modeling of fluid-structure interaction (FSI):

The procedure includes comprehensive fluidic dynamic analysis to evaluate the aerodynamics forces induced to the sensor by pressure fluctuations, which are utilized as input to finite element transient simulations.

Displacement Vs time half-layer 2, center hole



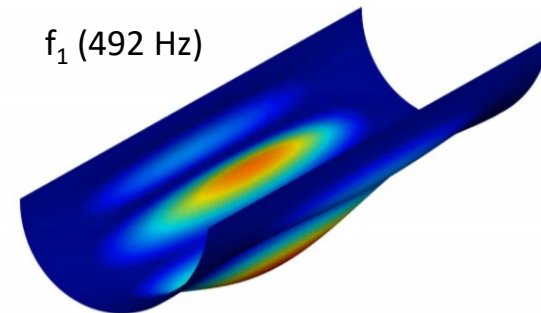
Power spectral density of the displacement for half-layer 2, $v_\infty = 8 \text{ m/s}$



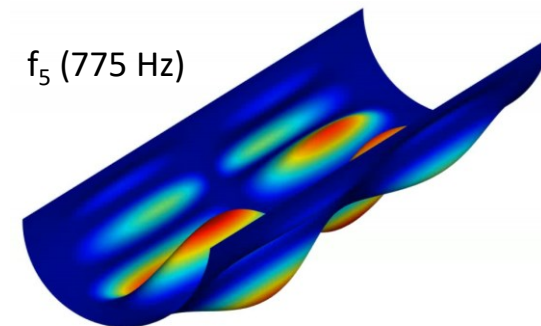
Displacement: Peak-to-peak $\sim 1.1 \mu\text{m}$, Root Mean Square $< 0.4 \mu\text{m}$

Modal analysis H-L2

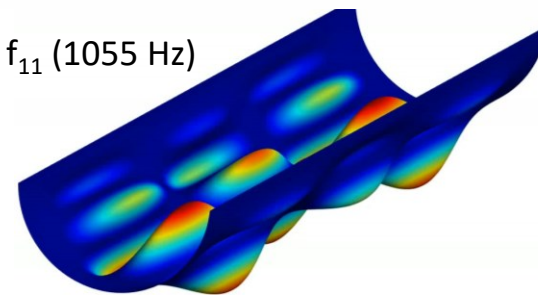
f_1 (492 Hz)



f_5 (775 Hz)



f_{11} (1055 Hz)



Testing: Thermoelastic failure assessment



No structural damage in a range of 10÷50°C and thermal peak up to 45 °C .

Prototype

BBM5

Made of final-grade materials



Thermal cycles

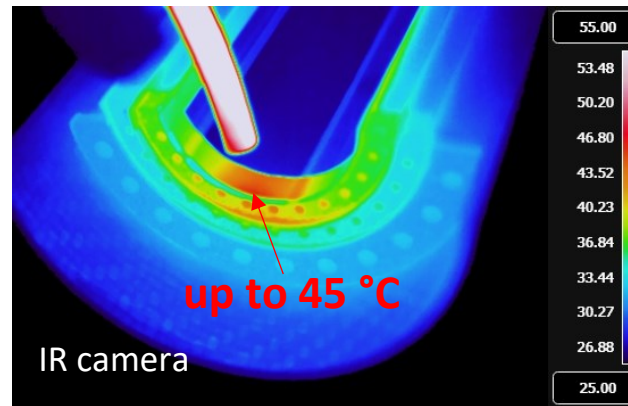
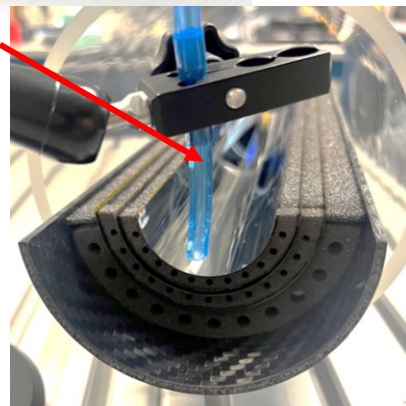
10 °C ÷ 50 °C

Hot-spot testing

- Localized heating to assess the effect of thermal gradient

Heat gun

Hot spot
up to 45 °C



Thermal cycles

@EP-DT QART lab
Climate chamber



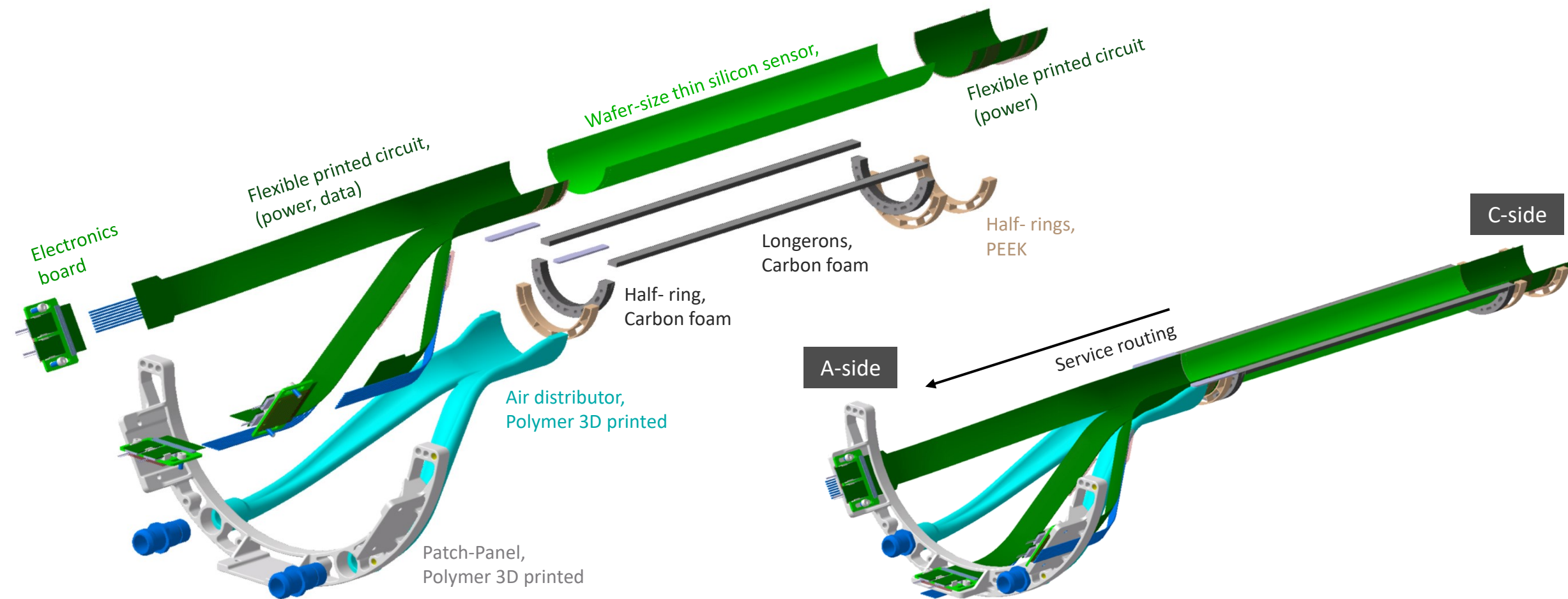
Humidity 30-40%

Services: half-layer



Challenge: integrate power/data lines and cooling ducts in minimum space, use of specific FPC design and 3d printing

CAD model of Half-layer 0: (left) Exploded view and (right) assembled view.

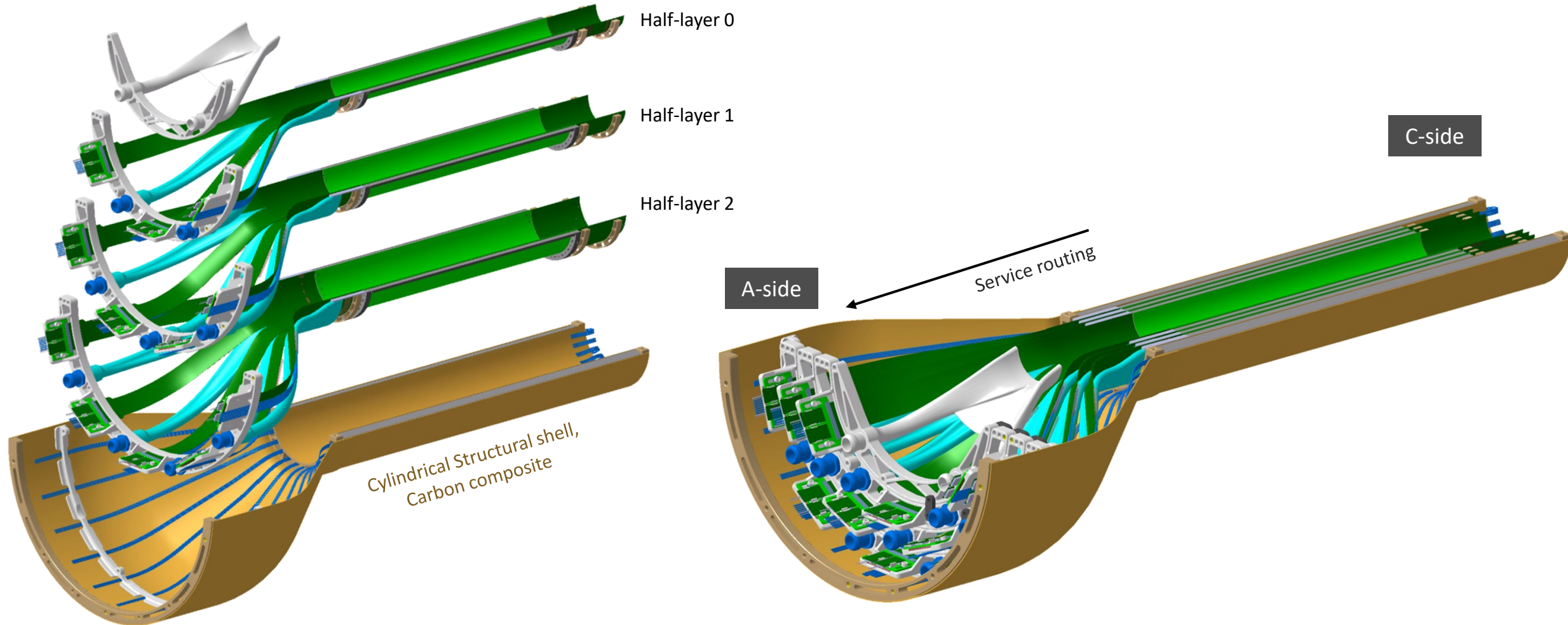


Services: half-detector



Challenge: handling and precise positioning and integration of layers and services inside the mechanical exoskeleton

CAD model of Half-layer 0: (left) Exploded view and (right) assembled view.

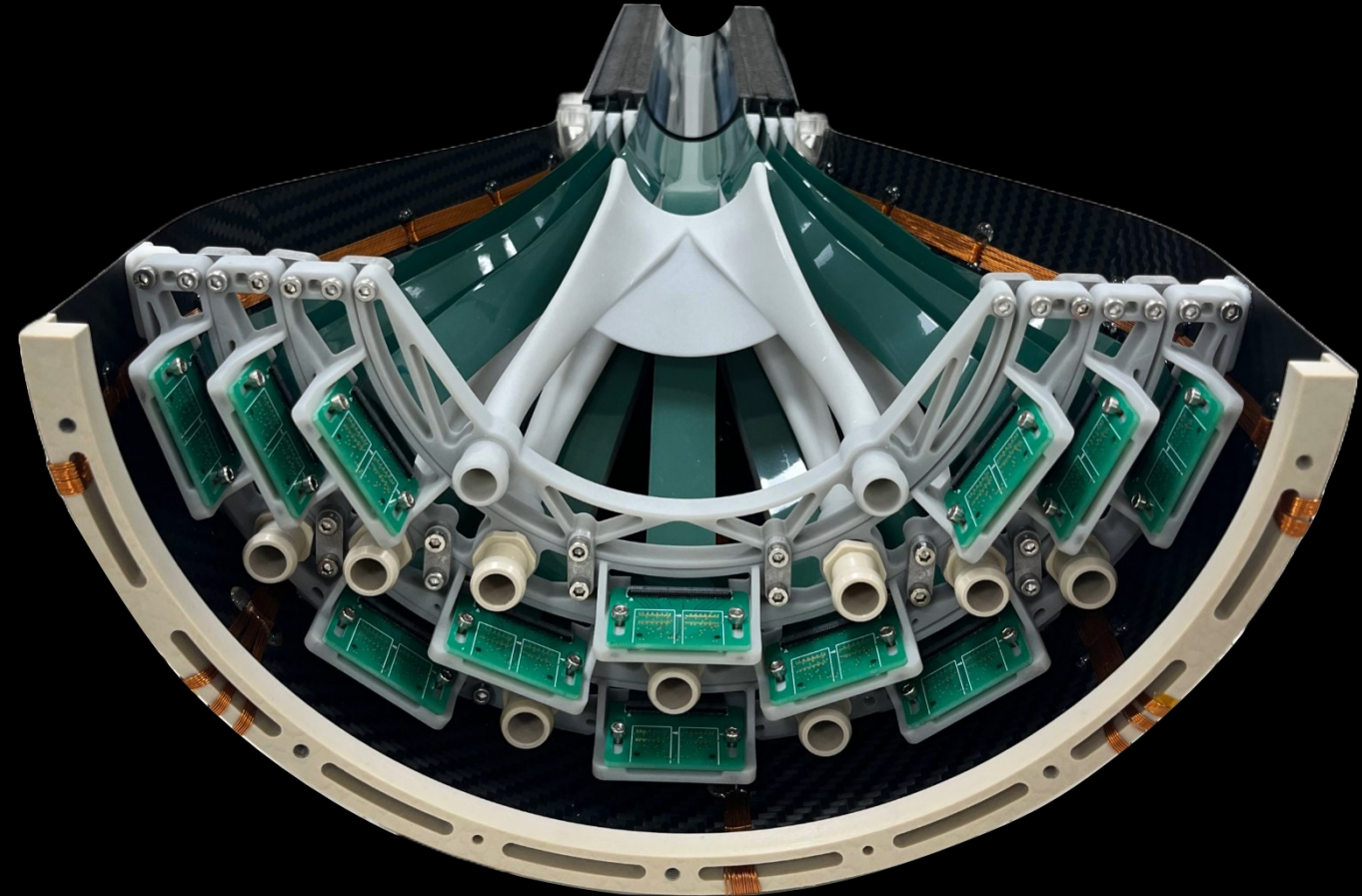
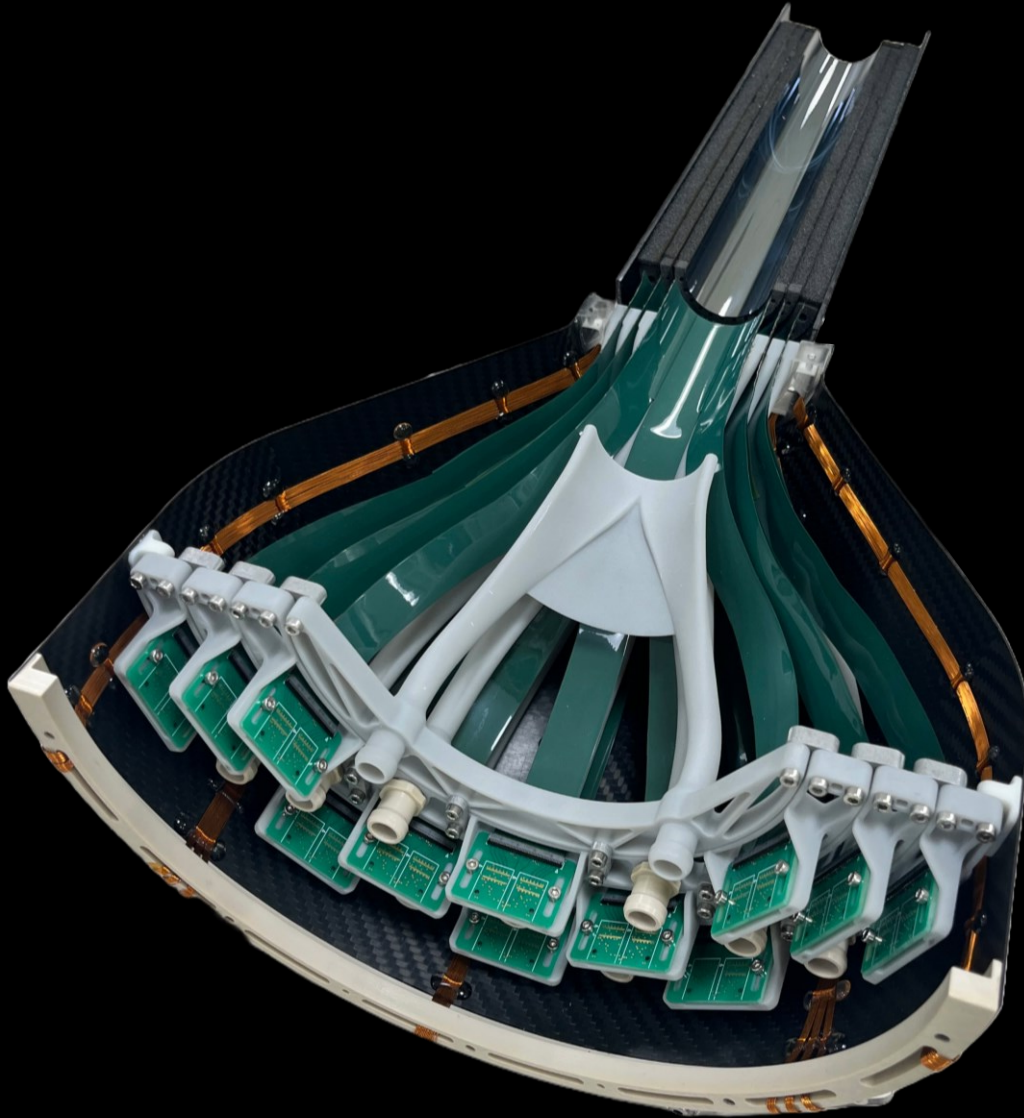


Power cables from C-side

Services: prototyping



The challenge here involves **finding space** for all the services and making them accessible and removable during assembly





- Wafer-size thin sensors successfully bent to cylindrical shape to form the detector's layer
- Air-cooling based on carbon foam radiator developed and satisfying thermal and stability requirements
- Layer connected to front end electronics by wire bonding
- Three layers integrated in a half barrel layout
- Service design implemented
- Different models built for the design validation
- **Next:** build a final-quality half-barrel (QM) within 2025 to be ready for final detector assembly (FM) in 2026

